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# An Algorithm to Estimate Unsteady and Quasi-Steady Pressure Fields from Velocity Field measurements

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**An algorithm to estimate unsteady and quasi-steady pressure fields from velocity field measurements**  4 John O. Dabiri<sup>1\*</sup>, Sanjeeb Bose<sup>2</sup>, Brad J. Gemmell<sup>3</sup>, Sean P. Colin<sup>4</sup>, and John H. Costello<sup>5</sup> *<sup>1</sup> Graduate Aeronautical Laboratories and Bioengineering, California Institute of Technology, Pasadena, California 91125, USA <sup>2</sup> Center for Turbulence Research, Stanford University, Stanford, California 94305, USA <sup>3</sup> Whitman Center, Marine Biological Laboratory, Woods Hole, Massachusetts 02543, USA <sup>4</sup> Marine Biology and Environmental Sciences, Roger Williams University, Bristol, Rhode Island 02809, USA <sup>5</sup> Biology Department, Providence College, Providence, Rhode Island 02918, USA*  14 \*Author for correspondence (email: jodabiri@caltech.edu) 16 Short title: Pressure estimation from velocity fields 18 Keywords: swimming, flying, wakes, feeding, particle image velocimetry 

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#### 22 **Summary**

23 We describe and characterize a method for estimating the pressure field corresponding to 24 velocity field measurements, such as those obtained by using particle image velocimetry. The 25 pressure gradient is estimated from a time series of velocity fields for unsteady calculations or 26 from a single velocity field for quasi-steady calculations. The corresponding pressure field is 27 determined based on median polling of several integration paths through the pressure gradient 28 field in order to reduce the effect of measurement errors that accumulate along individual 29 integration paths. Integration paths are restricted to the nodes of the measured velocity field, 30 thereby eliminating the need for measurement interpolation during this step and significantly 31 reducing the computational cost of the algorithm relative to previous approaches. The method is 32 validated by using numerically-simulated flow past a stationary, two-dimensional bluff body and 33 a computational model of a three-dimensional, self-propelled anguilliform swimmer to study the 34 effects of spatial and temporal resolution, domain size, signal-to-noise ratio, and out of plane 35 effects. Particle image velocimetry measurements of a freely-swimming jellyfish medusa and a 36 freely-swimming lamprey are analyzed using the method to demonstrate the efficacy of the 37 approach when applied to empirical data.

#### 40 **Introduction**

41 A longstanding challenge for empirical observations of fluid flow is the inability to 42 directly access the instantaneous pressure field using techniques analogous to those established 43 to measure the velocity field. Recent approaches have made significant progress, especially in 44 the measurement of pressure associated with unsteady fluid-structure interactions (e.g. Hong and 45 Altman, 2008; Jardin et al. 2009a, 2009b; David et al., 2009; Rival et al. 2010a, 2010b; David et 46 al., 2012; Tronchin et al., 2012; van Oudheusden, 2013; Liu and Katz, 2013). However, prior 47 efforts have not achieved explicit pressure estimation for moving bodies with time-dependent 48 shape, such as those characteristic of animal locomotion and feeding. The pressure field of 49 swimming animals is complicated by the interaction between pressure associated with vortices in 50 the flow and the irrotational pressure field due to acceleration of the body, often referred to as the 51 acceleration reaction or added mass (Daniel, 1984).

38 39

62

53 Existing methods for empirical pressure estimation often require relatively complex 54 measurement techniques such as multi-camera or time-staggered, multi-exposure particle image 55 velocimetry (Jensen and Pedersen, 2004; Liu and Katz, 2006). In addition, significant 56 computational costs can be associated with the post-processing required to derive the pressure 57 field from measurements of the velocity or acceleration fields. These post-processing approaches 58 generally fall into one of two categories. In the first case, the pressure field is computed as a 59 solution to a Poisson equation, e.g. in an inviscid flow:

 $\nabla^2 p = -\rho \left( \nabla \cdot \frac{\partial \mathbf{u}}{\partial t} \right)$  $\overline{a}$  $\overline{a}$ 61  $\nabla^2 p = -\rho \nabla \cdot \frac{D\mathbf{u}}{D}$ , (1)

63 where *p* is the pressure, **u** is the velocity vector, ρ is the fluid density, and *D*/*Dt* is the material 64 derivative, i.e., the time rate of change of an idealized infinitesimal fluid particle in the flow. 65 Solution of equation (1) poses challenges in practice because measurement errors accumulate 66 due to the required temporal and spatial derivatives of **u**, the condition number (i.e. sensitivity) 67 of the Laplacian operator (Golub and Van Loan, 1996), and measurement uncertainty in the 68 boundary conditions, especially at fluid-solid interfaces (Gurka et al., 1999). For attached flows 69 at high Reynolds number, the Neumann boundary condition specifying the pressure gradient at 70 fluid-solid interfaces is given by the boundary layer approximation as  $\partial p / \partial n \approx 0$ , where *n* is the 71 direction of the local normal surface vector (Rosenhead, 1963). However, for separated flows at 72 moderate or low Reynolds numbers, such as those commonly found in animal locomotion, *a*  73 *priori* determination of the appropriate fluid-solid boundary conditions for solution of (1) can be 74 intractable.

75 A second category of approaches for pressure field estimation is those based on direct 76 integration of the pressure gradient term in the Navier-Stokes equation, e.g. for incompressible 77 flow:

78

$$
\nabla p = -\rho \left( \frac{D \mathbf{u}}{Dt} - \nu \nabla^2 \mathbf{u} \right)
$$
 (2)

81 where ν is the kinematic viscosity of the fluid. The pressure difference between two points in the 82 domain is determined by integration of equation (2) between the two points. For example, the 83 difference in pressure between two points  $\mathbf{x}_1$  and  $\mathbf{x}_2$  is given by

84

85 
$$
p_2 - p_1 = \int_{x_1}^{x_2} \nabla p d\mathbf{x}
$$
 (3)

86

87 Because measurement errors accumulate along the path of integration from  $\mathbf{x}_1$  to  $\mathbf{x}_2$  in equation 88 (3), various techniques have been employed to make this approach less sensitive to measurement 89 uncertainty. A common strategy is to take advantage of the scalar property of the pressure field, 90 such that its local value is independent of integration path. Therefore, each independent 91 integration path that arrives at a point in the flow is in principle an independent estimate of the 92 pressure at that point, provided that measurement errors are uncorrelated. By polling a large 93 number of integration paths, an estimate of the local pressure can be achieved. For example, one 94 successful method (Liu and Katz, 2006) uses an iterative scheme that averages 2*m*(*n*+*m*) + 95 2*n*(2*m*+*n*) integration paths on an *m* x *n* grid in order to estimate the instantaneous pressure field.

96 The aforementioned iterative scheme, while effective in limiting the influence of 97 measurement errors, still incurs a relatively high computational cost. For example, for a 128 x 28 grid of velocity vectors that is commonly acquired using PIV, the method requires 1.6 x  $10^5$ 99 integration paths per iteration of velocity field integration; and several iterations can be required 100 for convergence of the method (Liu and Katz, 2006). Furthermore, if each integration path is 101 taken as a straight line through the domain, then the method requires interpolation of the 102 estimated pressure gradient field in order to evaluate integration path points that do not coincide 103 with the original data grid. While these requirements are not necessarily prohibitive for two-104 dimensional calculations, they are time-consuming and are indeed a showstopper for extension 105 of the method to three dimensions.

106 We present a simple yet demonstrably effective approach for pressure estimation that is 107 in the spirit of the second category of pressure estimation methods. The method is validated by 108 using two numerically-simulated flows: flow past a two-dimensional, stationary bluff body and 109 the flow created by a three-dimensional, self-propelled anguilliform swimmer. The first flow is 110 used to characterize a quasi-steady implementation of the algorithm, in which the pressure field

111 is estimated from a single velocity field measurement. The second flow demonstrates the ability 112 of the method to accurately estimate the pressure on unsteady, deformable bodies such as those 113 of relevance in animal locomotion. Both flows are used to characterize the method, including its 114 numerical convergence properties and sensitivity to domain size, signal-to-noise ratio, and out of 115 plane effects. Furthermore, we apply the method to PIV measurements of a freely-swimming 116 jellyfish medusa and a freely-swimming lamprey, showing that this tool can be applied to the 117 type of measurement data commonly acquired in research.

118

#### 119 **Materials and Methods**

120 *Material acceleration estimation* 

121 The instantaneous fluid particle acceleration *D***u**/*Dt* required for calculation of the 122 pressure gradient in equation (2) is estimated by advecting idealized infinitesimal fluid particles 123 in the measured velocity fields. For quasi-steady estimation, the material acceleration is derived 124 from a single velocity field as

125

127

126 
$$
\frac{D\mathbf{u}}{Dt}(\mathbf{x}_i) \approx \frac{\mathbf{u}_{PIV}(\mathbf{x}_i - \mathbf{u}_{PIV}(\mathbf{x}_i))}{\Delta t}
$$
(4)

128 where  $i = 1, 2...$  *m* x *n* (i.e. the dimensions of the velocity grid),  $\mathbf{x}_i$  are the positions of fluid 129 particles coincident with the grid points in the PIV velocity field, and  $\mathbf{x}^a_i$  are the positions of 130 those fluid particles after being advected by the instantaneous velocity field for a period Δ*t*: 131

132 
$$
\mathbf{x}_{i}^{a} \approx \mathbf{x}_{i} + \mathbf{u}_{PIV}(\mathbf{x}_{i})\Delta t
$$
 (5)

133

134 In order for equations (4) and (5) to remain valid, Δ*t* is limited to values much smaller 135 than the characteristic time scale of the flow, yet sufficiently large that there is a measurable 136 change in the fluid particle velocity.

137 For many flows, especially those involving accelerating or deforming bodies, the 138 aforementioned constraint on Δ*t* cannot be satisfied. For these inherently unsteady fluid-structure 139 interactions, we derive the material acceleration from two sequential velocity fields as 140

$$
141 \qquad \frac{D\mathbf{u}}{Dt}(\mathbf{x}_i, t_1) \approx \frac{\mathbf{u}_{PIV}\left(\mathbf{x}_i^{a}, t_2\right) - \mathbf{u}_{PIV}\left(\mathbf{x}_i, t_1\right)}{t_2 - t_1} \tag{6}
$$

142 where 
$$
\mathbf{x}_i^a \approx \mathbf{x}_i + \left(\frac{\mathbf{u}_{PV}(\mathbf{x}_i, t_1) + \mathbf{u}_{PV}(\mathbf{x}_i, t_2)}{2}\right)(t_2 - t_1)
$$
 (7)

144 Equation (7) is akin to a Crank-Nicolson (i.e. trapezoidal) scheme for the particle positions, in 145 contrast to the forward Euler scheme in equation (5). Hence, the convergence of the method with 146 time step is second order (Crank and Nicolson, 1947).

147 The primary source of measurement error in this type of unsteady estimate of the material 148 acceleration *D***u**/*Dt* arises from temporal noise in the measured velocity components at each node 149 in the velocity field. We address this by applying a temporal filter to the time series of velocity 150 fields, which results in a smoothing spline approximation **u**\* to the velocity **u** at each node in the 151 velocity field. The spline approximations are defined such that they minimize, for each 152 component of **u**, the parameter

153

154 
$$
S_{\mathbf{u}} = \phi \sum_{\tau=1}^{N} (\mathbf{u}_{\tau} - \mathbf{u}_{\tau}^*)^2 + (1 - \phi) \int_{t_{\text{min}}}^{t_{\text{max}}} \left( \frac{d^2 \mathbf{u}^*}{dt^2} \right)^2 dt
$$
 (8)

155

156 where  $\tau = 1...N$  is the temporal sequence of velocity fields to be analyzed,  $\mathbf{u}_\tau$  is a velocity vector 157 corresponding to velocity field  $\tau$  in the sequence,  $\mathbf{u}_\tau$  \* is the spline-approximated value of the 158 same velocity vector for the same velocity field in the sequence,  $t_{\text{min}}$  and  $t_{\text{max}}$  are the temporal 159 bounds on the sequence of velocity fields, and φ is a weight between the first and second terms 160 and has a value between 0 and 1. In effect, the parameter *S***u** quantifies both the deviation of the 161 spline approximation from the original data (i.e. the first term) and the total curvature magnitude 162 of the spline approximation (i.e. the second term). For  $\phi = 0$ , only the second term is minimized, 163 resulting in a least-squares fit with zero curvature, i.e. a linear fit to the data. For  $\phi = 1$ , only the 164 first term is minimized, giving a cubic spline fit that passes through each original data point. In 165 all that follows, we set  $\phi = 0.05$ , a value we have identified as enabling effective temporal noise 166 filtering without discarding true temporal trends in the measurement data.

168 is shown that the use of the temporal filter increases the order of temporal convergence above 169 second order, as anticipated by theory (Atkinson, 1968).

170 It is worth noting that the distinction between the quasi-steady and unsteady approaches 171 can be made explicit by decomposing the material acceleration into its Eulerian components: 172

167 Further characterization of the temporal filter is provided in Appendix 2. In particular, it

173 
$$
\frac{D\mathbf{u}}{Dt} = \frac{\partial \mathbf{u}}{\partial t} + (\mathbf{u} \cdot \nabla)\mathbf{u}
$$
 (9)

174

175 The quasi-steady approximation in equations (4) and (5) implicitly neglects the first term on the 176 right-hand side of equation (9), whereas the unsteady calculation retains it.

177 The viscous term on the right-hand side of equation (2) is computed using centered finite 178 differences between adjacent nodes in the velocity field. The effect of the viscous term is 179 evaluated in the context of a numerical simulation described in the validation section.

## 180

#### 181 *Pressure gradient integration*

182 Whereas previous methods that integrate the pressure gradient via many integration paths 183 assign to each grid point the arithmetic mean of the many integrations, in the present approach 184 the paths are polled by taking the median. The median is less sensitive to grossly erroneous 185 values that may arise on a few of the integration paths due to localized measurement errors or 186 due to localized errors created by the aforementioned material acceleration approximations in 187 equations (4) through (7). Hence, this approach enables a significant reduction in the total 188 number of integration paths per frame that are required to achieve accurate pressure estimates. 189 Fig. 1 illustrates the paths used presently. Eight families of integration paths are used, with each 190 family originating at the domain boundary and propagating toward each grid point from the left 191 (L), upper left (UL), top (T), upper right (UR), right (R), lower right (LR), bottom (B), and lower 192 left (LL), respectively.

193 Only 8 integration paths (one per family) per grid point are used, for a total of 8*m* x *n* 194 paths per velocity field. For the aforementioned example grid of 128 x 128 velocity vectors, 1.3  $195 \times 10^5$  integration paths are required, a 20 percent reduction from existing optimal methods (Liu 196 and Katz, 2006). More importantly, the integration paths are constrained to include only grid

197 points coincident with the original velocity field. For example, the UL integration path is 198 comprised of alternating integration steps in the -y and +x directions, originating at the domain 199 boundary and terminating at each grid point. Hence, no interpolation is required in order to 200 integrate the pressure gradient field. Furthermore, portions of many of the paths are redundant, 201 facilitating fast calculation using simple matrix manipulations. A forward Euler spatial 202 integration scheme is used throughout, resulting in first-order spatial convergence of the method 203 (see Appendix 1).

204 An important limitation of the present algorithm that arises from the trade-off between 205 speed and accuracy is that it assumes the pressure is zero at the point on the outer domain 206 boundary where each integration path is initiated. This does not imply, however, that the final 207 pressure estimate is constrained to be zero at the boundaries. Integration paths that originate from 208 the other domain boundaries and terminate at a given boundary may estimate a non-zero value of 209 pressure at the termination point. If the median of all paths terminating at that point on the 210 domain boundary is non-zero, then the final pressure estimate at that point will also be non-zero. 211 Note that for all points in the domain, the final pressure estimate is relative to a zero reference 212 pressure, as that is the pressure at the origin of each integration path. The impact of these 213 assumptions on the robustness of the technique is quantified below, and it is shown to be modest 214 for the external flows tested. At the same time, the net result of this tradeoff in the algorithm 215 design is a more than order-of-magnitude reduction in computational time compared to previous 216 methods (see Appendix 2).

217 A common source of localized error that can affect pressure estimates is the presence of 218 solid objects in the flow. Typical PIV measurements are often unreliable in the region close to 219 solid objects, which compromises pressure integration paths that cross the fluid-solid interface, 220 especially in previous methods that average the erroneous data instead of discarding it via 221 median polling (or in Poisson solvers that rely on the pressure gradient at the fluid-solid interface 222 as a boundary condition). In the present algorithm, integration paths that cross a fluid-solid 223 interface in the flow can be nullified by assigning the nodes nearest to the interface an undefined 224 pressure gradient. Hence, when that value is integrated along any integration path, the pressure 225 value for that path also becomes undefined and therefore does not contribute to the median 226 calculation.

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#### 228 *Validation data sets*

229 To validate the accuracy of the quasi-steady pressure estimates achieved using this 230 algorithm, a numerical simulation of flow past a two-dimensional square cross-section cylinder 231 at a Reynolds number of 100 was used. This numerical data set enabled quantification of the 232 effects of spatial resolution, domain size, and signal-to-noise ratio, while providing a known 233 pressure field standard for comparison (see Appendix 1). The numerical simulation was 234 executed using a solver that computes on arbitrary polyhedra (Ham and Iaccarino, 2004). In the 235 present case, a regular Cartesian mesh was utilized and subsequently interpolated onto coarser 236 grids of varying sizes typical of PIV data. The viscous term in equation (2) was retained in all of 237 the calculations to demonstrate the robustness of the median polling approach to errors normally 238 associated with application of the Laplacian operator. For all calculations of equations (4) and 239 (5) in this validation, we set  $\Delta t = 0.01h/U_{max}$ , where *h* is the mean grid spacing and  $U_{max}$  is the 240 maximum flow speed in the measurement domain. The results described below were insensitive 241 to order of magnitude larger and smaller values of Δ*t*. Where noted, nearest-neighbor Gaussian 242 smoothing was applied both to the pressure gradient before integration and to the resulting 243 pressure field.

244 The accuracy of the fully unsteady pressure estimates was validated by using a published 245 numerical simulation of a three-dimensional, self-propelled anguilliform swimmer (Kern and 246 Koumoutsakos, 2006). The Reynolds number based on swimmer length and speed was 247 approximately 2400. Time steps between sequential velocity fields from 0.02*T* to 0.08*T* (where *T* 248 is the swimming stroke duration) were studied to quantify the temporal convergence of the 249 method. The validation results described below are based on calculations of equations (6) and (7) 250 using velocity fields separated by 0.02*T*.

251

## 252 *Empirical data sets*

253 The present method was also applied to particle image velocimetry measurements of a 254 freely-swimming *Aurelia aurita* (Linnaeus, 1758) jellyfish medusa and a freely-swimming 255 *Anguilla rostrata* (LeSueur, 1817) lamprey to demonstrate the performance of the algorithm with 256 empirical data inputs and, in the case of the jellyfish, without treatment of fluid-solid interfaces. 257 The swimming Reynolds numbers of the jellyfish and lamprey were approximately 1000 and 258 10,000, respectively, and the time between sequential velocity fields was 5 ms ( $t/T \approx 0.013$ ) and

259 4 ms ( $t/T \approx 0.015$ ), respectively. Details of the PIV implementation can be found in published 260 literature (Colin et al., 2012).

261

## 262 **Results and discussion**

### 263 *Quasi-steady pressure estimation*

264 Fig. 2 compares an instantaneous pressure field from the numerical simulation of flow 265 past a stationary bluff body with the pressure field estimated from the corresponding velocity 266 field using the quasi-steady form of the present algorithm. A vector field spatial resolution of 267 *D*/16 (where *D* is the side length of the bluff body) is used in the horizontal and vertical 268 directions to mimic typical PIV measurements. The salient features of the flow, especially the 269 high pressure on the upstream face of the bluff body and the low pressure in the shear layers and 270 near-wake vortices, are well captured by the algorithm (see Appendix 1 for discussion of 271 discrepancies in the far wake). Furthermore, the maximum and minimum pressures in the field 272 are in quantitative agreement (Fig. A1). To be sure, nearest-neighbor Gaussian smoothing creates 273 a spurious thin layer of undefined pressure at the fluid-solid interface and moves the pressure 274 peak on the upstream face of the body away from the interface. However, the correct near-body 275 pressure can be recovered by increasing the grid resolution so that the nearest-neighbor filter 276 artifact on the body surface is limited to a smaller region very close to the body. Additional 277 surface pressure calculations for the quasi-steady case (Appendix 1) are based on a velocity 278 vector spacing of *D*/64. Note that a similar increase in resolution using a PIV camera would 279 require a concomitant reduction in the measurement window size due to limits on camera pixel 280 resolution.

281 Additional characterization of the quasi-steady algorithm is detailed in Appendix 1, 282 including analysis of spatial convergence; the relative contribution of each integration path to the 283 median pressure field; robustness to measurement noise; and the effects of domain size, fluid 284 viscosity, and fluid-solid interfaces.

285

## 286 *Unsteady pressure estimation*

287 Fig. 3 compares an instantaneous pressure field from the numerical simulation of a self-288 propelled anguilliform swimmer with the pressure field estimated from the corresponding 289 velocity field using the unsteady form of the present algorithm. A vector field spatial resolution

The Journal of Experimental Biology – ACCEPTED AUTHOR MANUSCRIPT The Journal of Experimental Biology - ACCEPTED AUTHOR MANUSCRIPT 290 of *L*/42 (where *L* is the length of the swimmer) is used in the horizontal and vertical directions. 291 No smoothing is applied to this data set in order to contrast the results with those in the previous 292 section and to limit the spatial extent of the region of undefined pressure near the fluid-solid 293 interface. The algorithm is effective in capturing the high-low pressure couples formed on the 294 sides of the swimmer head and tail as they accelerate in the positive-*y* direction; the low-high 295 pressure couple formed at the mid-body as it accelerates in the negative-*y* direction; and the 296 pressure in the wake vortices.

297 The importance of the unsteady term in equation (9) is illustrated by comparison with the 298 pressure field estimated using the quasi-steady approximation, shown in Fig. 3C. Low pressure 299 in the wake vortices is captured, but the high-low pressure couples on the body surface due to the 300 body added mass are missing entirely, as is the high pressure in the wake due to vortex added 301 mass (Dabiri, 2006). The comparison is further quantified in Fig. 4, which plots the pressure on a 302 contour surrounding the swimmer and immediately adjacent to the region of undefined pressure. 303 At each of the four phases of the swimming cycle shown, good agreement is achieved between 304 the pressure computed in the numerical simulation and the pressure estimated from the velocity 305 field using the unsteady algorithm. By contrast, the pressure estimated by the quasi-steady 306 algorithm is erroneous everywhere except near the forming wake vortex at the tail.

307 Additional characterization of the unsteady algorithm is provided in Appendix 2, 308 including analysis of temporal convergence and out-of-plane effects for three-dimensional flows.

309 To demonstrate the efficacy of the present method for analyzing empirically measured 310 velocity fields, Fig. 5 shows measured velocity and vorticity fields for the freely-swimming 311 jellyfish and lamprey (panels A and C) along with the corresponding pressure fields estimated 312 using the unsteady algorithm (panels B and D). The full measurement domain is shown in both 313 cases; the velocity vector field is plotted at half of the full resolution. Only the left half of the 314 jellyfish body is visible in the measurement domain; its exumbrellar surface is indicated by a 315 black curve in panels A and B. The full lamprey body is visible in panels C and D.

316 In both cases, the pressure field derived from the velocity field measurements captures 317 key features near the body surface and in the wake. In particular, the jellyfish data set indicates 318 low pressure in the forming starting vortex and high pressure where the bell margin is 319 accelerating inward and pushing the adjacent fluid. The results are consistent with the measured 320 vorticity field (panel C), with the region of low pressure corresponding to the core of the starting 321 vortex. The presence of low and high pressure regions near the bell margin is also in agreement 322 with previous numerical simulations of a swimming jellyfish with similar body shape and 323 kinematics (Sahin et al., 2009).

324 The lamprey data set shares similarities with the three-dimensional numerical model 325 shown previously. The vorticity and pressure fields are less smooth and show finer structure in 326 the empirical measurements, which is attributable in part to the Reynolds number being 327 approximately four times higher than that of the numerical simulation.

328 The ease of implementation of this algorithm, both in terms of data acquisition and 329 velocity field post-processing, and its relatively low computation cost (see Appendix 2) gives it 330 the potential to find use in a broad range of problems of interest in biological fluid mechanics. 331 Because the temporal filter implemented in the unsteady algorithm does add considerable time to 332 the pressure calculation (cf. Fig. A10), in practice one should first evaluate the results of both the 333 quasi-steady and the fully unsteady implementations of the algorithm on a sample of the data of 334 interest to determine whether unsteady effects are important. If they are not, then the quasi-335 steady calculation provides the most efficient tool for determination of the pressure field.

336 Although the present evaluation focused on two-dimensional velocity fields, it is 337 straightforward to extend the algorithm to three dimensions by the addition of a limited number 338 of new integration paths consistent with the geometry in Fig. 1. In that case, even greater 339 reductions in computation expense can be achieved relative to existing methods due to the 340 relatively small total number of required integration paths and the elimination of associated 341 velocity field interpolation during integration of the pressure gradients.

342 A free MATLAB implementation of this algorithm is available at 343 http://dabiri.caltech.edu/software.html.

344

#### 345 **Appendix 1: Additional Characterization of Quasi-steady Algorithm**

## 346 *Effect of median polling*

347 To illustrate the contribution of each integration path to the final pressure estimate, Fig. 348 A1 plots the pressure on the body surface (at 0.1*D* away from the fluid-solid interface, to avoid 349 the spatial filter artifact) and on two additional concentric square contours in the domain (e.g. the 350 dashed contour in Fig. A2A), as computed using each of the 8 families of integration paths. The 351 results illustrate the benefit of median polling versus an average of the integration path results.

352 For example, only 5 of the integration path families intersect the upstream face of the bluff body 353 without passing through the body itself. The median of these curves is in good quantitative 354 agreement with the correct surface pressure (Fig. A1A). The pressure profiles for the two 355 concentric square contours in the domain (i.e. Figs. A1B and A1C) indicate that the contribution 356 of each family to the final pressure estimate is spatially non-uniform. This is illustrated 357 qualitatively in Fig. A1D, which is a contour plot that colors each point in the domain according 358 to the path family that corresponds to the median pressure at that point. Because there are 8 path 359 families, the median is always the average of the two intermediate values (where none of the 360 paths is undefined due to intersection with the solid body). To reveal the individual integration 361 path family contributions, a ninth pressure value equal to the mean of the 8 path families is 362 included in Fig. A1D, so that the median pressure is from either a single integration path family 363 or from the mean. The contour plot indicates that each integration path family contributes to the 364 final pressure field estimate, but the contributions are often spatially localized. The pressure 365 estimates for the R family of integration paths are noticeably less accurate than the other families 366 (e.g. Fig. A1A) and yet, as illustrated in Fig. A1D, these paths determine the pressure estimate in 367 the far wake. This leads to the observed poorer pressure estimate in that region of the flow (e.g. 368 Fig. 2B). The underlying source of this effect is discussed below in the section examining the 369 effect of boundary conditions.

## 371 *Effect of global measurement error*

372 Perhaps the most important test of the algorithm is its robustness to global measurement 373 errors, such as those associated with empirical measurements. Fig. A2 illustrates the streamwise 374 velocity contours for data sets with increasing levels of Gaussian white noise superimposed on 375 the *u* and *v* velocity components. The highest levels of noise, corresponding to the lowest signal-376 to-noise ratios, are higher than typical PIV data but possibly representative of instantaneous two-377 dimensional data collected in a highly-turbulent flow field, where out-of-plane motion can 378 reduce data quality. Comparison of the pressure profiles on a square contour centered on the 379 bluff body and with side length 3*D* so that it passes through the salient flow features (i.e. Fig. 380 A2A) indicates that, with the exception of the highest noise level tested, the quantitative pressure 381 estimates remain consistent with the noise-free result despite relatively high noise (Fig. A3A). 382 Error in the pressure estimate is not additive with the increasing noise level because

383 errors do not accumulate uniformly on the 8 paths that arrive at each point in the domain. Hence, 384 median polling remains an effective filter irrespective of the noise magnitude, up to the second-385 highest noise level tested. At higher noise levels, contour plots of the pressure estimate begin to 386 exhibit spatial discontinuities reminiscent of the median contributions in Fig. A1D. Because the 387 pressure estimates from each integration path family begin to diverge in the presence of high 388 noise levels, median polling in this case leads to spatially discontinuous changes in pressure. 389 Result of this sort are an indication that measurement noise in the input velocity data has become 390 unacceptably large.

#### 392 *Effect of boundary conditions*

393 As mentioned previously, a major assumption implicit in the present algorithm is that the 394 pressure on each integration path is zero at its originating point on the boundary, to avoid the 395 need for a computationally expensive iteration scheme to solve for the boundary pressure as part 396 of the field solution (Liu and Katz, 2006). Although this assumption can be reasonable for large 397 domains, it is prudent to investigate the dependence of the pressure estimate on the domain size. 398 Fig. A3B plots the pressure on a square contour centered on the bluff body (see Fig. A2A) for 399 domains ranging in size from *H*/*D* = 2 to 30, where *H* is the half-width of the domain. The results 400 indicate that the accuracy of the algorithm (and hence, the assumption regarding the boundary 401 pressure) is not significantly compromised until the domain shrinks to  $H/D = 2$ . This limitation is 402 important to keep in mind when designing experiments that will make use of the present 403 algorithm.

404 Notwithstanding the demonstrated efficacy of the aforementioned assumption regarding 405 the boundary pressure, examination of the individual pressure estimates on each family of 406 integration paths reveals that some individual estimates are severely compromised by this 407 assumption. Most notably, the R family of integration paths originate at the downstream 408 boundary of the domain, where vortices shed by the bluff body exit the measurement window 409 and create a non-zero pressure on that boundary. Hence this family of pressure estimates is 410 significantly less accurate than the others, as seen in Fig. A1A for example. The benefit of the 411 median polling approach is that this estimate is usually discarded in determining the final 412 pressure estimate. In contrast, previous methods would include pressure estimates affected by the 413 downstream boundary in the final averaged pressure estimate, and therefore require additional

414 computational effort to resolve the correct pressure on that boundary via iterative processes.

415 However, the present method does suffer in that the pressure in regions close to the 416 downstream boundary is based either on integration paths that originate at the downstream 417 boundary where the pressure is nonzero (i.e. R, UR, and LR families) or on long integration 418 paths from the other boundaries. The relatively large error accumulated on the long integration 419 paths can make them an even poorer estimate of the local pressure near the R boundary (cf. Fig. 420 A1D); hence the median pressure in this region is less accurate than in the rest of the domain. 421 This limitation is inherent in the present method and should be kept in mind when using the 422 technique for flows with large velocity gradients at any of the boundaries.

## 423

#### 424 *Effect of fluid viscosity*

425 It is useful to examine the role of the viscous term in equation (2), as many previous 426 pressure estimation methods neglect this term. Fig. A4A plots the pressure estimates on the body 427 surface for each integration path family as in Fig. A1A, but for a pressure estimate that neglects 428 the viscous term in the Navier-Stokes equation. The effect is most noticeable in integration paths 429 orthogonal to the mean flow (i.e. T and B), especially near the upstream face of the bluff body. 430 This result can be understood by considering the contributions to the pressure gradient from the 431 streamwise and transverse material acceleration components, *Du*/*Dt* and *Dv*/*Dt*, relative to the 432 contributions from the Laplacian of the streamwise and transverse velocity components in the 433 viscous term. As the flow approaches the upstream face of the bluff body, the material 434 acceleration is dominated by streamwise fluid particle deceleration *Du*/*Dt*. However, the pressure 435 computed on integration paths that are orthogonal to the mean flow (i.e. T and B) is independent 436 of this term. Instead, on these paths the pressure depends to a significant degree on the local 437 velocity curvature (i.e. second spatial derivative) as the flow is turned around the bluff body. 438 This effect is captured in part by the Laplacian of the transverse velocity. Hence, its neglect leads 439 to an underestimate of the pressure on those integration paths. The net effect of the neglected 440 viscous term is minimal due to the median polling approach implemented presently, i.e., the T 441 and B paths do not represent the median pressure estimate on the upstream face of the bluff body 442 and are therefore not a factor in the final pressure estimate in that region of the flow. 443

#### 444 *Effect of fluid-solid interfaces*

445 An aspect of the present algorithm that can be cumbersome is the treatment of the fluid-446 solid interface to eliminate the effect of integration paths that pass through solid objects in the 447 flow. For example, for moving objects, this approach requires the identification of the fluid-solid 448 interface in each data frame. To illustrate the effect of the interface treatment in the algorithm, 449 Fig. A4B plots pressure estimates on the body surface, where the algorithm has been 450 implemented without treating the fluid-solid interface. The accuracy of the algorithm is 451 noticeably affected due to additional spurious pressure estimates on paths that cross the body. 452 However, it is noteworthy that the final pressure estimate is still qualitatively consistent with the 453 correct pressure field. It may therefore be acceptable to bypass the fluid-solid interface treatment 454 where only a quantitative approximation of the pressure field is sought. The results of the 455 analysis in Fig. 5B, which did not identify the boundary of the medusa as was done for the 456 numerical data, suggest that suitable pressure estimates can still be achieved where treatment of 457 the fluid-solid interfaces is not practical.

#### 459 *Spatial convergence*

460 The spatial convergence of the quasi-steady algorithm was evaluated by computing the 461 pressure on a square contour immediately adjacent to the region of undefined pressure on the 462 bluff blody, and by integrating the pressure to compute the net force in the streamwise and lateral 463 directions. Fig. A5 plots the fractional error in these calculations (using the pressure from the numerical simulation (CFD) as the true value, i.e.  $|F_{CFD} - F_{estimate}|/F_{CFD}$ ) versus the grid 465 resolution. At relatively large grid spacing, the log-log curve has a slope of 1, indicating the 466 expected first-order spatial convergence of the method. As the grid spacing is further reduced, 467 the error decreases more slowly. This effect is due to a combination of inherent model error and 468 numerical round-off error. A convergence plot for calculations of the time-averaged streamwise 469 force is included. Its departure from first order convergence at small grid spacing confirms that 470 the quasi-steady approximation is not solely responsible for errors at small grid spacing. For grid 471 spacing less than *D*/16, the error falls below 5 percent for the instantaneous pressure and 472 approaches 10 percent for the time-averaged pressure.

473

#### 474 **Appendix 2: Additional Characterization of Unsteady Algorithm**

#### 475 *Effect of temporal filter*

476 Fig. A6 plots the time series of *v* component data at two selected points in the jellyfish 477 PIV data set. Despite the relatively smooth spatial distribution of velocity, as illustrated in Fig. 478 5A and Fig. A7A, there is non-trivial scatter in the temporal data at both spatial locations. Flow 479 accelerations computed by using finite differences of adjacent velocity fields would be subject to 480 large errors because the local slope varies considerably between adjacent pairs of velocity fields. 481 A temporal filter of the data is therefore essential in this case. Fig. A6 indicates the 482 corresponding smoothing splines that were fit to the data and subsequently used to compute the 483 material acceleration. The splines capture the true transient behavior of the flow while 484 eliminating the high-frequency noise. Comparison of Fig. A7A and A7B illustrates that the 485 spatial distribution of velocity is relatively unaffected by the temporal filter. It is prudent to note 486 that if a flow exhibits real, high-frequency oscillations in the velocity, e.g. in turbulence, it will 487 be essential that the PIV measurements are of sufficiently high temporal resolution such that the 488 smoothing spline does not discard those temporal trends. In those cases, it is important that the 489 frequency of PIV measurements satisfies the Nyquist sampling criterion with respect to the time 490 scale of turbulence fluctuations, while concurrently avoiding sampling at frequencies high 491 enough to incur numerical round-off errors in the calculation (Beckwith et al., 2007).

#### 493 *Effect of out-of-plane flow*

494 Given that two-dimensional PIV data represents a projection of three-dimensional flow, it 495 is useful to characterize the impact of that limitation on the accuracy of the present methods. As 496 in prior work (Stamhuis and Videler, 1995), Fig. A8 characterizes the out-of-plane motion by 497 computing the divergence of a two-dimensional velocity field extracted from the three-498 dimensional numerical simulation of the self-propelled swimmer and of the PIV data sets 499 examined in Fig. 5. The divergence is made dimensionless by multiplying it by the time step 500 between adjacent velocity fields, as this time scale is most relevant for calculation of the material 501 acceleration. The plots effectively quantify the fractional change in the volume of an idealized 502 infinitesimal fluid particle between adjacent velocity fields. Because the flows are 503 incompressible, the fractional volume change would be identically zero if the flows were two-

504 dimensional. Deviation from zero values can therefore be attributed to velocity gradients 505 perpendicular to the plane of the velocity field, i.e., out-of-plane flow.

506 The results in Fig. A8 indicate that the three-dimensional numerical simulation exhibits 507 greater out-of-plane flow than the PIV measurements. Given the demonstrated accuracy of the 508 algorithm in the case of the three-dimensional numerical data, we can conclude that the 509 algorithm is robust to out-of-plane effects at the magnitudes found in typical PIV measurements. 510 To be sure, the divergence metric does not capture out-of-plane flow where there is no flow 511 gradient in the perpendicular direction. However, in such cases, the PIV would itself be difficult 512 to acquire, as the seed particles would not remain in the plane of the laser sheet sufficiently long 513 to enable temporal cross-correlation of their positions.

#### 515 *Temporal convergence*

516 The temporal convergence of the unsteady algorithm was evaluated by plotting the 517 fractional error in the pressure at the head of the self-propelled swimmer at an instant of high 518 acceleration (using the pressure from the numerical simulation (CFD) as the true value, i.e. 519  $\left| p_{\text{CFD}} - p_{\text{extimate}} \right| / p_{\text{CFD}}$  ) versus the time step between velocity fields (Fig. A9). Although the 520 available data was limited to time steps from 0.02*T* to 0.08*T*, the results are consistent with 521 temporal convergence that is higher than second order, except as the smallest step size is 522 approached, where further reduction in error is limited by inherent model error and numerical 523 round-off error. At a temporal spacing of 0.02*T*, the error in the pressure at the head is 524 approximately 8 percent.

525 When the unsteady algorithm is applied to a sequence of velocity fields that are spaced 526 too closely in time, leading to increased numerical error, the results appear similar to those 527 described in the context of global measurement error (Appendix 1) in which the pressure 528 contours exhibit spatial discontinuity reminiscent of Fig. A1D.

529

## 530 *Computational cost*

531 Fig. A10 plots the time required for a single 3-GHz processor to apply the temporal filter 532 and to compute the pressure field for velocity fields from 32x32 to 256x256 nodes, which is a 533 practical upper limit for typical PIV measurements due to camera pixel resolution. The time 534 required for the temporal filter scales linearly with the number of nodes in the velocity field. The

535 cost is independent of the number of velocity fields in the sequence of up to the tested several 536 hundred frames of data. The computational cost of the subsequent pressure calculation scales 537 sublinearly in the range tested, and it is significantly lower than the cost of the temporal filter in 538 absolute terms. For example, for a 128x128 velocity field, each pressure field is computed in 539 approximately 3 seconds, as compared to 46 seconds using an existing iterative algorithm (Liu 540 and Katz, 2006).

543 **Acknowledgments** 

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607 **Figure 1.** Geometry of integration paths for pressure field estimation. Eight paths originate from 608 the domain boundary and propagate toward each point  $(x_i, y_i)$  in the domain from the left  $(L)$ , 609 upper left (UL), top (T), upper right (UR), right (R), lower right (LR), bottom (B), and lower left 610 (LL). The points on each path coincide with the measurement grid.



613 **Figure 2.** (A) Pressure field computed from numerical simulation of flow past two-dimensional square cylinder at a Reynolds number Re =  $UD/v = 100$ . The pressure coefficient  $c_p = p/(\rho U^2)$ . 615 (B) Pressure field estimated using quasi-steady algorithm.



618 **Figure 3.** (A) Pressure field computed from numerical simulation of three-dimensional self-619 propelled swimmer. The pressure coefficient  $c_p = p/(\rho U^2)$ . Velocity nodes completely inside 620 swimmer body are indicated in black (body surface is smooth in numerical simulation). Spatial 621 coordinates are normalized by swimmer length. (B) Pressure field estimated using unsteady 622 algorithm. (C) Pressure field estimated using quasi-steady algorithm. 623



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625 **Figure 4.** Pressure on contour surrounding the self-propelled swimmer and immediately adjacent 626 to the region of undefined pressure, at four instants during the swimming cycle duration *T***.** Head 627 is at body node number 45; tail is at body nodes 1 and 90. Solid curve, pressure computed from 628 numerical simulation. Closed circles, pressure estimated from unsteady algorithm. Open circles, 629 pressure estimated from quasi-steady algorithm. (A)  $t/T = 1/4$ . (B)  $t/T = 1/2$ . (C)  $t/T = 3/4$ . (D) 630  $t/T = 1$ .



632 **Figure 5.** (A) Particle image velocimetry (PIV) measurement of a freely-swimming jellyfish 633 medusa. Velocity field is plotted on vorticity contours. Maximum velocity vector is 634 approximately 3 cm  $s^{-1}$ . Velocity field is plotted at half of full resolution. Left half of 635 exumbrellar surface is indicated by black curve. (B) Pressure field estimated using unsteady 636 algorithm. (C) Particle image velocimetry (PIV) measurement of a freely-swimming lamprey. 637 Velocity field is plotted on vorticity contours. Maximum velocity vector is approximately 11 cm  $638 \,$  s<sup>-1</sup>. Velocity field is plotted at half of full resolution. Animal body is approximately indicated in 639 black. (D) Pressure field estimated using unsteady algorithm.

640



642 **Figure A1.** (A) Pressure on surface of bluff body estimated using quasi-steady algorithm. 643 Measurement contour is offset by 0.1*D* from the fluid-solid interface. The pressure coefficient *cp*  $\epsilon = p/(\rho U^2)$ . *s* is the local surface coordinate and increases in the counter-clockwise direction from 645 the upper right corner of the bluff body. Dashed black line, pressure from numerical simulation; 646 solid black line, pressure estimated using quasi-steady algorithm; solid colored lines, pressure 647 estimates based on each family of integration paths. Colors correspond to paths in Fig. 1 and to 648 the legend in panel D. (B, C) Pressure on square contours centered on the bluff body and with 649 side length 2*D* and 3*D*, respectively (e.g. Fig. A2A). *s* is the local surface coordinate and 650 increases in the counter-clockwise direction from the upper right corner of each square contour. 651 The difference in abscissa in panels A-C reflects the different contour perimeters. (D) Contour 652 plot that colors each point in the domain according to the path family that corresponds to the 653 median pressure at that point. To reveal the individual integration path family contributions, a 654 ninth pressure value equal to the mean of the 8 path families is included, so that the median 655 pressure is from either a single integration path family or from the mean.



657 **Figure A2.** Streamwise velocity contours for flow field with Gaussian white noise added to 658 reduce the signal-to-noise ratio (SNR). (A)  $SNR = 32$  dB. (B)  $SNR = 24$  dB. (C)  $SNR = 20$  dB. 659 (D) SNR = 16 dB. Dashed square in panel A indicates contour on which quasi-steady pressure 660 estimates are compared in Fig. A3A.



663 **Figure A3.** (A) Quasi-steady pressure estimate on contour shown in Fig. A2A for varying signal-664 to-noise ratio. *s* is the local surface coordinate and increases in the counter-clockwise direction 665 from the upper right corner of the square contour. (B) Quasi-steady pressure estimate on contour 666 shown in Fig. A2A for varying measurement domain size. *H* is the half-width of the 667 measurement domain.



670 **Figure A4.** (A) Pressure on surface of bluff body estimated using quasi-steady algorithm without 671 viscous term. Measurement contour is offset by 0.1*D* from the fluid-solid interface. *s* is the local 672 surface coordinate and increases in the counter-clockwise direction from the upper right corner 673 of the bluff body. Dashed black line, pressure from numerical simulation; solid black line, 674 pressure estimated using quasi-steady algorithm; solid colored lines, pressure estimates based on 675 each family of integration paths. (B) Pressure on surface of bluff body estimated using quasi-676 steady algorithm without treatment of fluid-solid interfaces to remove integration paths that pass 677 through the solid body.



679

680 **Figure A5.** Spatial convergence of the algorithm. Log-log plot of the fractional error in 681 instantaneous streamwise (closed circles), instantaneous lateral (open circles), and time-averaged 682 streamwise (closed squares) force coefficients versus grid resolution for numerical simulation of 683 two-dimensional flow past the bluff body. Solid line indicates a slope of 1 corresponding to first-684 order convergence. Deviation from first-order convergence at small grid resolution is due to a 685 combination of model error and numerical round-off error.



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687 **Figure A6.** Time series of *v* component data at two selected points in the jellyfish PIV data set. 688 Symbols indicate original PIV data at corresponding locations identified in Fig. A7. Solid curves 689 indicate respective smoothing splines.



692 **Figure A7.** (A) Contour plot of *v* component of original velocity measurement during middle of 693 jellyfish bell contraction. (B) Contour plot of *v* component temporal spline-filtered velocity 694 measurement during middle of jellyfish bell contraction. Location of animal is similar to that 695 indicated in Fig. 5A, although earlier in the bell contraction phase. Closed circle near bell margin 696 and open circle in wake indicate locations of temporal profiles in Fig. A6.





699 **Figure A8.** Contour plots of normalized two-dimensional divergence for (A) three-dimensional 700 numerical simulation of self-propelled swimmer, (B) PIV measurement of freely-swimming 701 jellyfish (cf. Fig. 5A), (C) PIV measurement of freely-swimming lamprey (cf. Fig. 5C). 702 Dimensional divergence is normalized by multiplying by the time step between sequential

703 velocity fields in each case.



704

705 **Figure A9.** Temporal convergence of the algorithm. Log-log plot of the fractional error in 706 pressure at the head of the simulated three-dimensional self-propelled swimmer versus time step 707 between velocity fields (closed circles). Solid line indicates a slope of 2 corresponding to 708 second-order convergence.



711 **Figure A10.** Computational cost of the algorithm, as quantified by the time required for a single 712 3-GHz processor to apply the temporal filter (open circles) and to compute the pressure field 713 (closed circles) for velocity fields from 32x32 to 256x256 nodes. Solid line indicates slope of 1.